



Aerodynamic centre of wing-fuselage-nacelle combinations: effect of rear-fuselage pylon-mounted nacelles

Associated software: ESDUpac A1309 See ESDU 13009



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THE PREPARATION OF THIS DATA ITEM

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Mr J.W.H. Thomas	 British Aerospace, Hatfield-Chester Division
Mr H. Vogel	 British Aerospace, Weybridge-Bristol Division.

* Corresponding Members

The work on this Item was carried out in the Aircraft Motion Group of the Engineering Sciences Data Unit under the supervision of Mr P.D. Chappell, Group Head. The member of staff who undertook the technical work involved in the initial assessment of the available information and the construction and subsequent development of the Item was

Mr R.W. Gilbey

— Senior Engineer.

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AERODYNAMIC CENTRE OF WING-FUSELAGE-NACELLE COMBINATIONS: EFFECT OF REAR-FUSELAGE PYLON-MOUNTED NACELLES

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AERODYNAMIC CENTRE OF WING-FUSELAGE-NACELLE COMBINATIONS: EFFECT OF REAR-FUSELAGE PYLON-MOUNTED NACELLES

1. NOTATION AND UNITS (see Sketch 1.1)

		SI	British
A	aspect ratio of equivalent-wing planform*		
а	lift-curve slope of equivalent-wing planform [*]	radian ⁻¹	radian ⁻¹
a _n	lift-curve slope of nacelle in isolation, based on area wl	radian ⁻¹	radian ⁻¹
<i>c</i> ₀	centre-line chord of equivalent-wing planform*	m	ft
$\overline{\overline{c}}$	aerodynamic mean chord of equivalent-wing planform *	m	ft
c _r	root chord of equivalent-wing planform [*]	m	ft
c _{ref}	general reference chord for stability calculations	m	ft
Н	downwash parameter (see Equation (2.3))	radian	radian
K	ratio of lift on nacelle-pylon-fuselage combination to lift on nacelle and pylon in isolation		
l	length of nacelle forward-cowl or overall length of single-cowl nacelle, see Figure 1	m	ft
М	Mach number		
т	fuselage length forward of leading edge of root chord of equivalent wing planform *	m	ft
n	fuselage length aft of trailing edge of root chord of equivalent-wing planform *	m	ft
r	distance of nacelle inlet aft of wing-fuselage aerodynamic centre	m	ft
r _{te}	distance of nacelle inlet aft of wing trailing-edge, measured from centre of nacelle inlet, see Sketch 1.1	m	ft
r'	distance of nacelle inlet aft of quarter-chord point of centre-line chord of equivalent-wing planform [*]	m	ft
r″	chordwise distance of nacelle lip (or inlet) from leading edge of aerodynamic mean chord of equivalent wing planform, measured positive aft, see Sketch 1.1	m	ft

For footnote refer to end of Notation.

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